

# Efficient uncertainty propagation through an aero-elastic wind turbine model

Juan P. Murcia: PhD. student, jumu@dtu.dk

Pierre-E. Réthoré: Supervisor, Senior Scientist

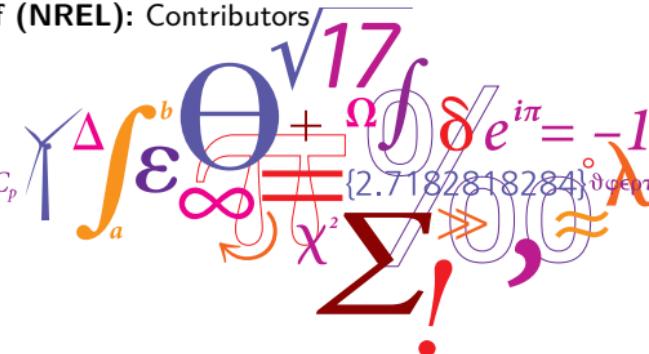
Anand Natarajan, John D. Sørensen: Co-supervisors

Nikolay Dimitrov, Taeseong Kim, Peter Graf (NREL): Contributors

Technical University of Denmark (DTU)

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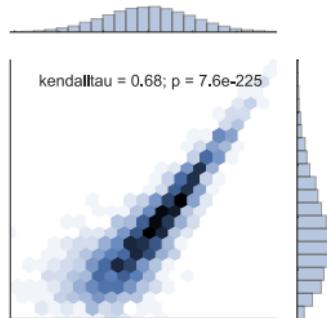
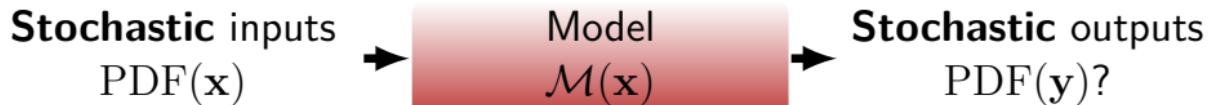
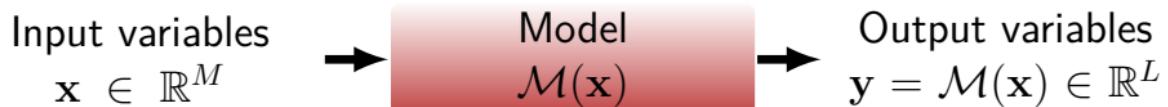
$$P = \frac{1}{2} \rho A v^3 C_p$$



# Outline

- ① (A very short) Introduction to Polynomial Chaos Expansions
- ② Polynomial Chaos Expansions as wind turbine aero-elastic model surrogates
- ③ Results
- ④ Future Work

# Uncertainty propagation problem



?

# Monte-Carlo simulation

- Sample the joint input distribution:

$$\mathbf{x}_i \sim \text{PDF}(\mathbf{x})$$

- Input sample can be generated using variance reduction methods such as: Latin Hypercube sampling (LHS), Halton or Hammersley sequences.
- Obtain a response sample by evaluating the model in each input realization:

$$\mathbf{y}_i = \mathcal{M}(\mathbf{x}_i)$$

- **Pros:** Very robust and easy to implement and parallelize
- **Cons:** Convergence is slow ( $\propto N^{-1/2}$ )

# Polynomial Chaos expansion

- Build a polynomial surrogate of the model:

$$y(\mathbf{x}) \approx \hat{y}(\mathbf{x}) = \sum_{l=0}^{N_c-1} c_l \phi_l(\mathbf{x})$$

- A polynomial basis,  $\phi_l(\mathbf{x})$ , is built with respect to  $\text{PDF}(\mathbf{x})$

Distribution	Polynomial Family
Uniform	Legendre
Normal	Hermite
Exponential	Laguerre

- The model is evaluated,  $\mathbf{y}_i = \mathcal{M}(\mathbf{x}_i)$ , and *projected/fitted* to the polynomial basis.
- A MC sample is generated using the surrogate.
- Sensitivity analysis is estimated from the MC sample Saltelli et al. [Saltelli 2010].
- **Pros:** Convergence is fast ( $\propto N^{-m}$ ,  $m > 1$ ,  $m$  is problem dependent)
- **Cons:** How to define the order of the polynomials in each variable? How to avoid over fitting? How to avoid oscillations?

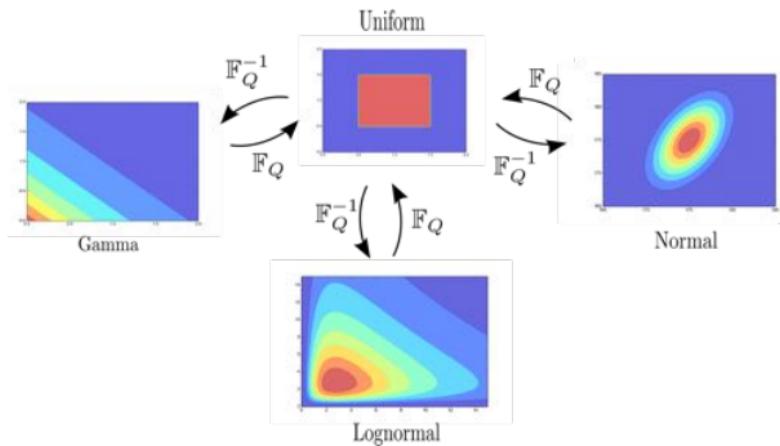
# How to deal with correlated inputs?

## Rosenblatt Transformation [Rosenblatt 1952]

- Transforms the correlated input variables ( $x$ ) into a multi-dimensional uncorrelated uniform space ( $w$ ). Solve the propagation problem in the uncorrelated space:

$$\mathbf{y}(\mathbf{x}) = \mathbf{y}(\mathbb{F}_Q^{-1}(\mathbf{w})) \approx \hat{\mathbf{y}}(\mathbf{w}) = \sum_{l=0}^{N_c-1} c_l \phi_l(\mathbf{w})$$

- Rosenblatt transformation consists in using the inverse of the CDF of each variable in sequence. Chaospy includes this transformation [Feinberg 2015]. Graph reproduced from Chaospy tutorials.



# Methods to find the coefficients $c_l$

## Semi-Spectral projection (quadrature integration)

- Use a quadrature rule to approximate the integrals (nodes,  $x_i$  and weights  $\omega_i$ ). Gaussian quadrature is widely used.

$$c_l = \langle y, \phi_l \rangle = \int y(x) \phi_l(x) \text{PDF}(x) dx \approx \sum_{i=0}^N \omega_i y(x_i) \phi_l(x_i)$$

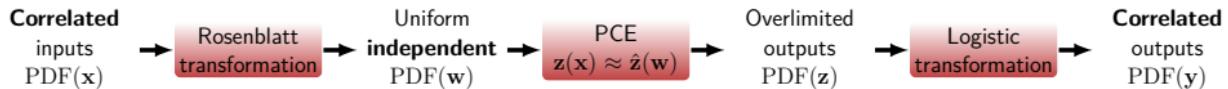
- **Pros:** Very good for low number of dimensions
- **Cons:** Unstable for heavy tailed PDFs. Quadrature rules fail with most correlated variables

## Point collocation (polynomial fit)

- Generate a small sample and fit the polynomial basis using Least squares or some other optimization method (e.g. LAR, LASSO).
- **Pros:** Very robust. Optimization algorithms are design to handle large number of dimensions (sparsity) and correlated inputs.
- **Cons:** Not as efficient as semi-spectral collocation.

# PCE of complex cases

## Variable transformations steps



### Rosenblatt transformation

- Used to decorrelate the variables
- Variables can be transformed to independent Uniform or Normal
- Inverse transformation used for MC sample. Use efficient sampling techniques in the unitary uniform uncorrelated space.

### PCE model surrogate

- Polynomial chaos expansion working on the uncorrelated space.
- Trained using k-Fold validation to avoid over-fitting and prefer lower order polynomials (Least absolute shrinkage and selection operator - LASSO problem).

### Logistic transformation

- Used to force fixed constraints in the outputs: i.e. to avoid overshoots.
- Can be used to smooth discontinuities and to impose only positive values.

# Polynomial Chaos Expansions as wind turbine aero-elastic model surrogates

# DTU 10 MW RWT

Inputs: 4D Conditionally Correlated.

Input variables

$$\mathbf{x} \in \mathbb{R}^4 \\ (\text{WS}; \sigma_1; \alpha; \gamma)$$



Model  
 $\mathcal{M}(\mathbf{x})$



Output variables  
 $\mathbf{y} = \mathcal{M}(\mathbf{x}) \in \mathbb{R}$

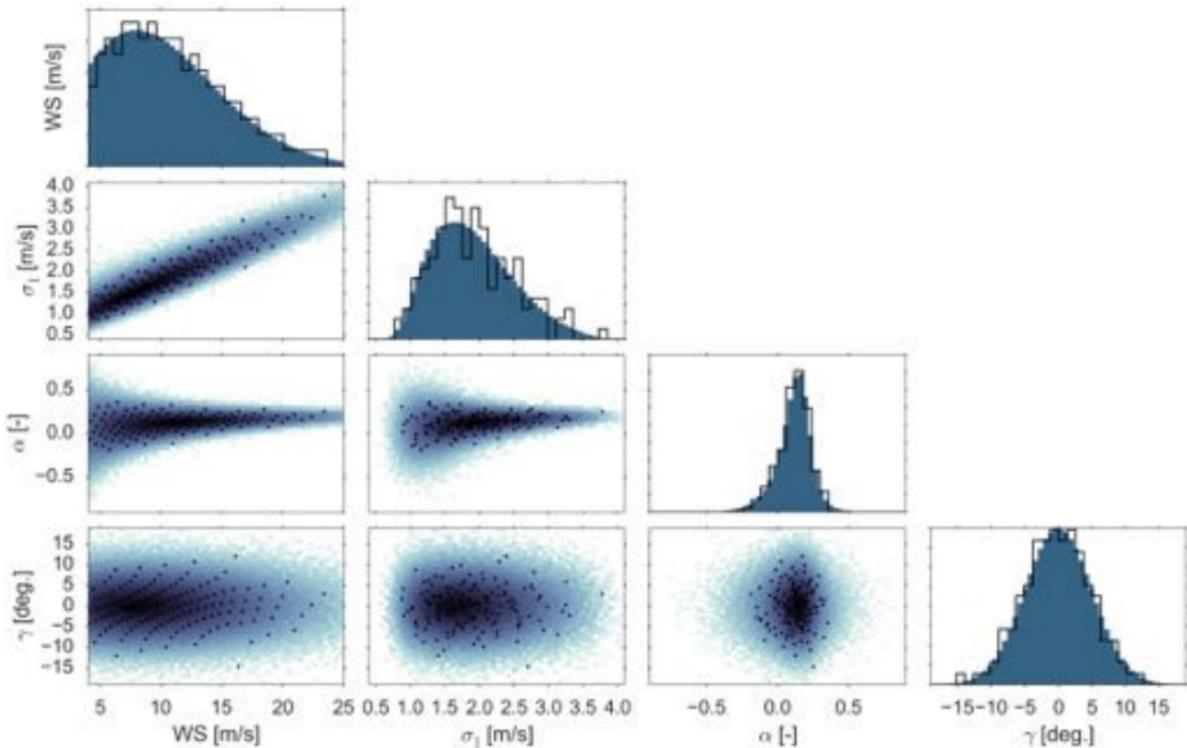
## Model

- DTU: 10 MW reference WT and HAWC2 with Mann turbulence.

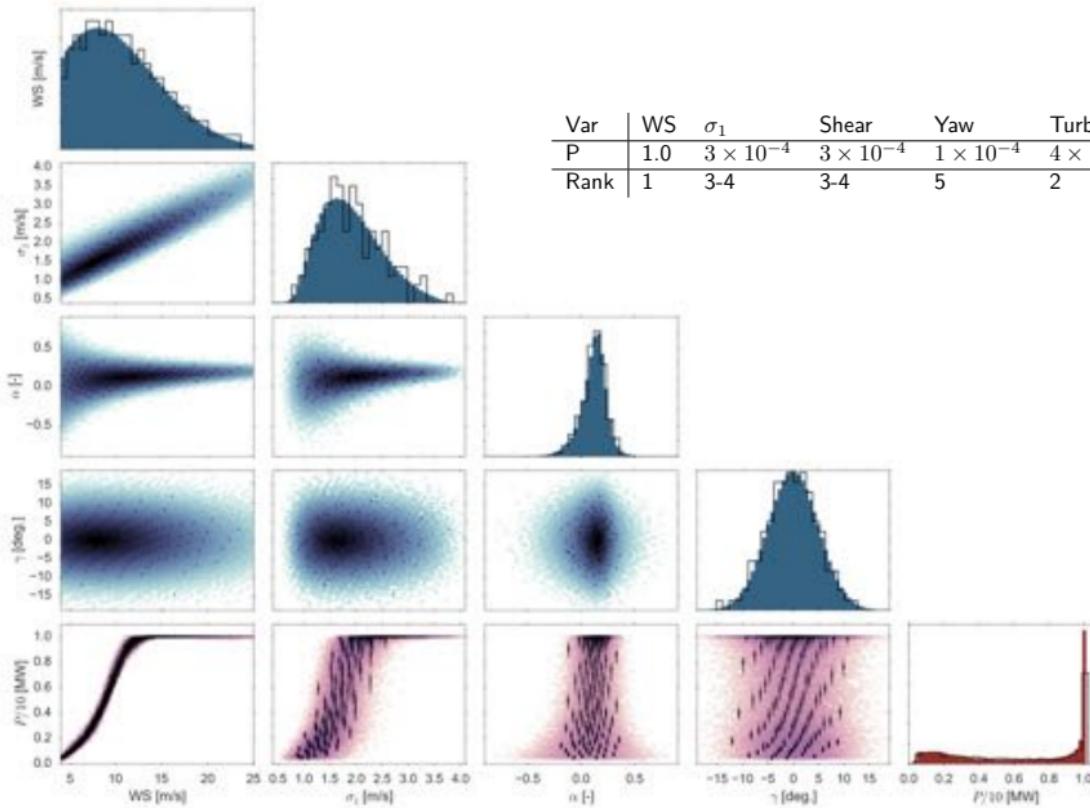
## Uncertain Input Distribution, Class I-A site.

- $\text{WS} \sim \text{Rayleigh}(\mu = 10)$
- $\text{TI}_{\text{ref}} = 16\%$
- $\sigma_1 \sim \text{Lognormal}(\mu = \mu(\text{WS}, \text{TI}_{\text{ref}}), \sigma = \sigma(\text{WS}, \text{TI}_{\text{ref}}))$ , NTM from IEC 61400-1.
- $\alpha \sim \text{Normal}(\mu = 0.088[\log(\text{WS}) - 1], \sigma = 1/\text{WS})$ , [Dimitrov 2015]
- $\gamma \sim \text{Normal}(0, \sigma = 5^\circ)$
- Obtain statistics from 100 TSeed realizations at each input

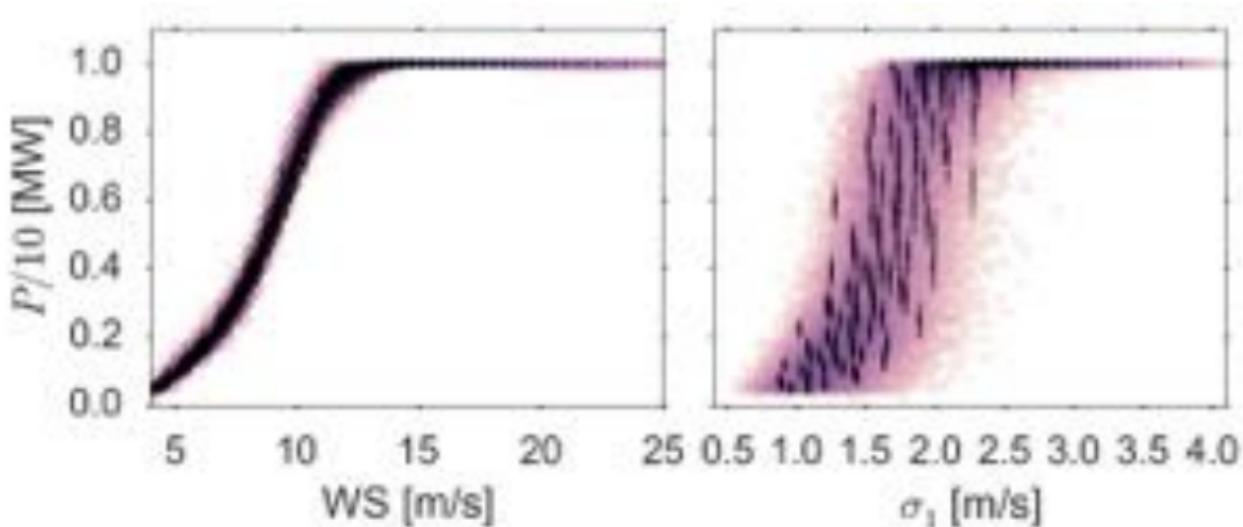
# DTU 10 MW RWT. Training sample



# DTU 10 MW RWT. Power surrogate



# DTU 10 MW RWT. Power surrogate



# How to deal with the turbulent inflow realization uncertainty?

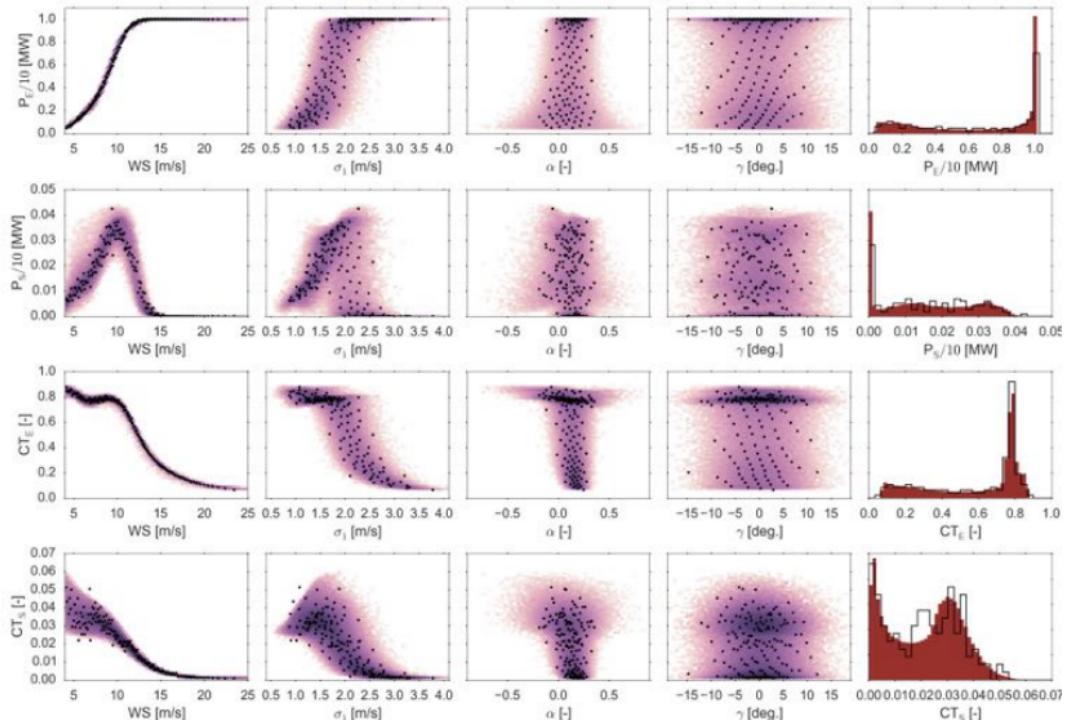
- Two polynomial chaos expansion are built independently for each output variable:

$$\hat{y}_{\mathbb{E}}(\mathbf{x}) \approx y_{\mathbb{E}}(\mathbf{x}) = \mathbb{E}(y|\mathbf{x}) \quad \hat{y}_{\mathbb{S}}(\mathbf{x}) \approx y_{\mathbb{S}}(\mathbf{x}) = \mathbb{S}(y|\mathbf{x})$$

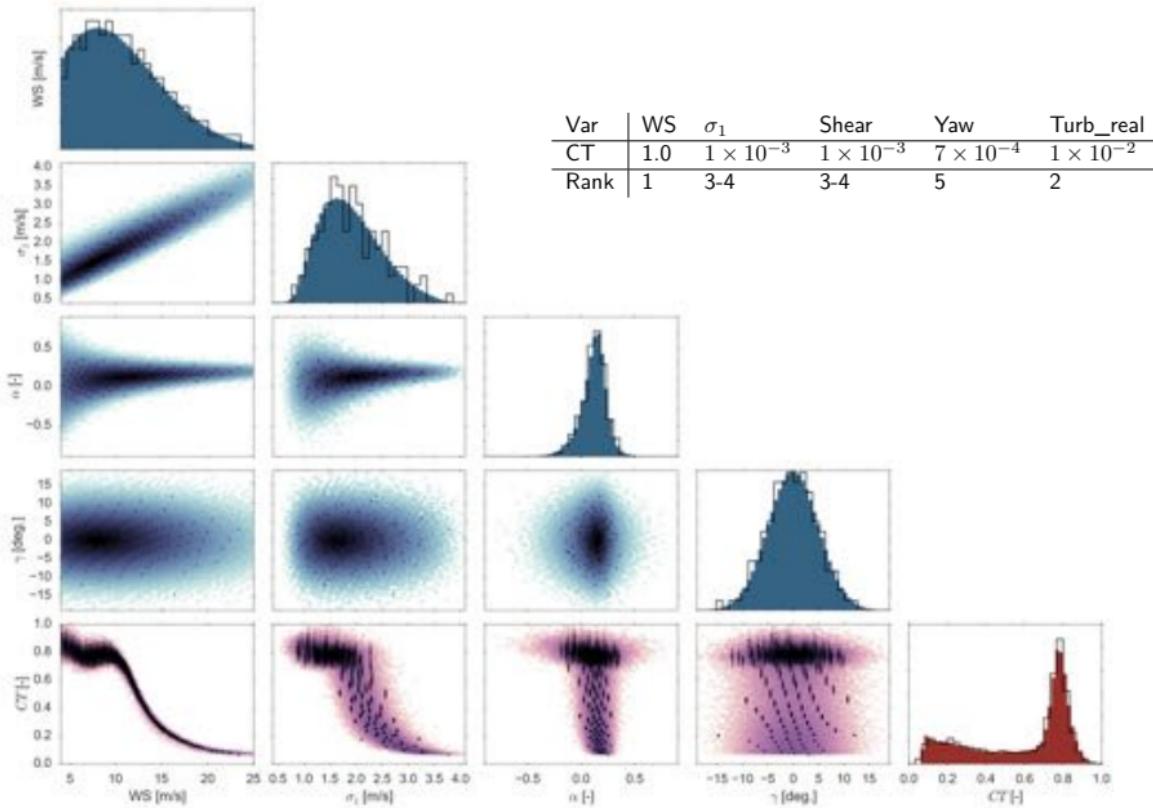
$$\hat{y}(\mathbf{x}) \sim \text{Normal}(\hat{y}_{\mathbb{E}}(\mathbf{x}), \hat{y}_{\mathbb{S}}(\mathbf{x}))$$

- $y_{\mathbb{S}}(\mathbf{x})$  is the local variation due to the different turbulent structures.
- For example  $P_{\mathbb{S}}$  represents the standard deviation of multiple 10-min averaged powers with different turbulent inflow realization.  $P_{\mathbb{S}}$  is NOT the standard deviation of the instantaneous power during the 10 minutes of simulation.

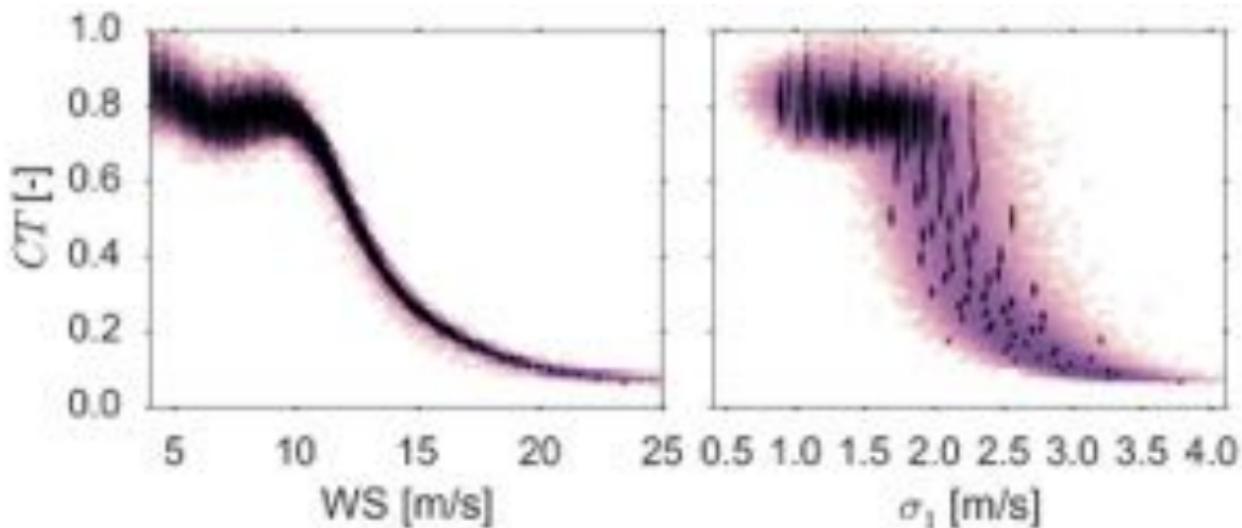
# DTU 10 MW RWT. PCE for individual local statistical moments



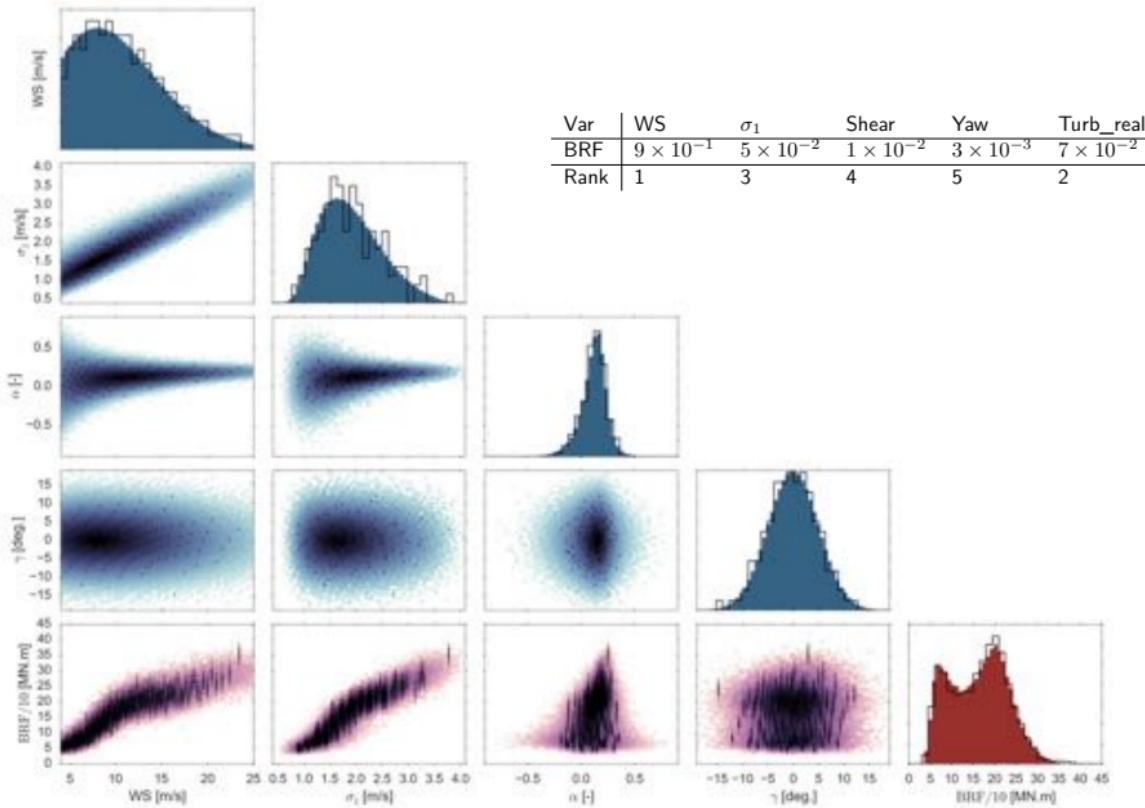
# DTU 10 MW RWT. Thrust coefficient surrogate



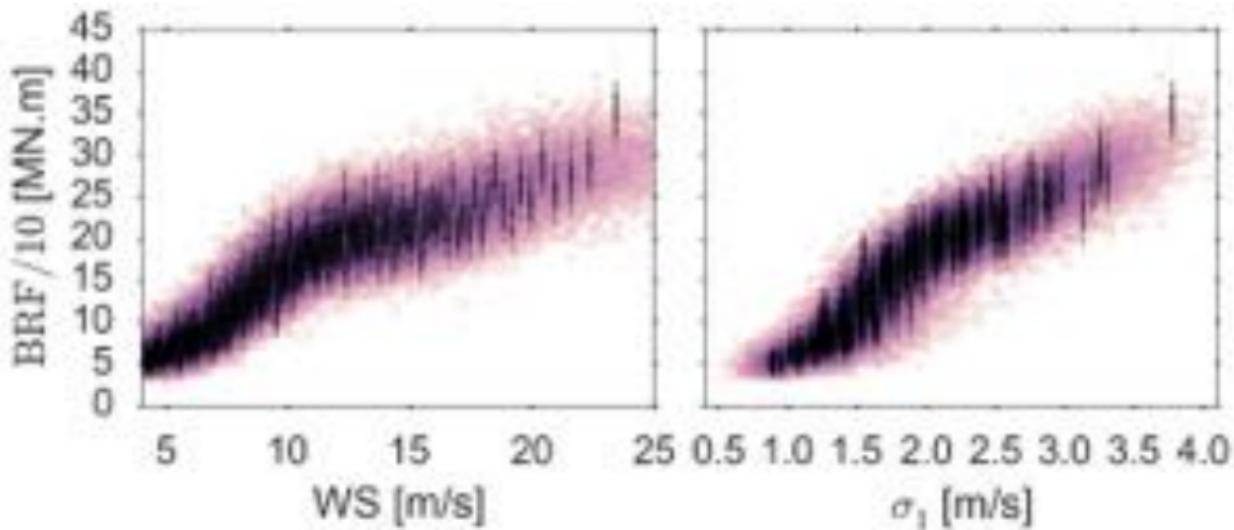
# DTU 10 MW RWT. Thrust coefficient surrogate



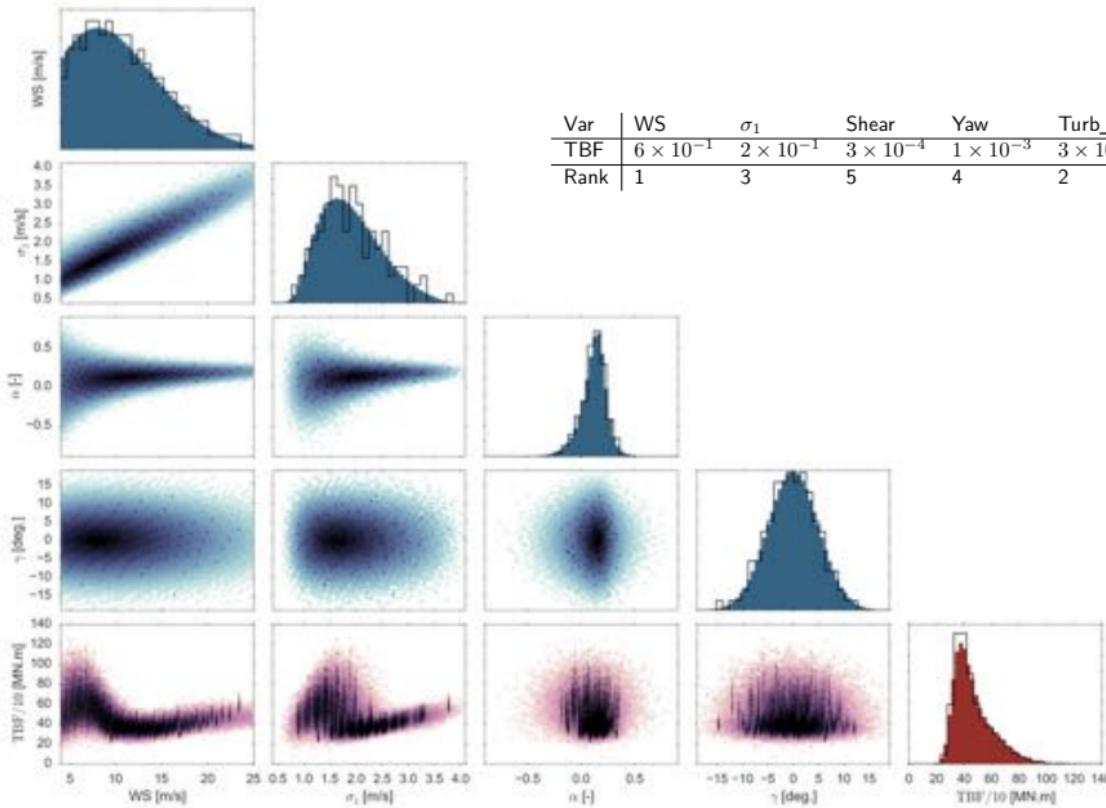
# DTU 10 MW RWT. Blade root flapwise EFL surrogate



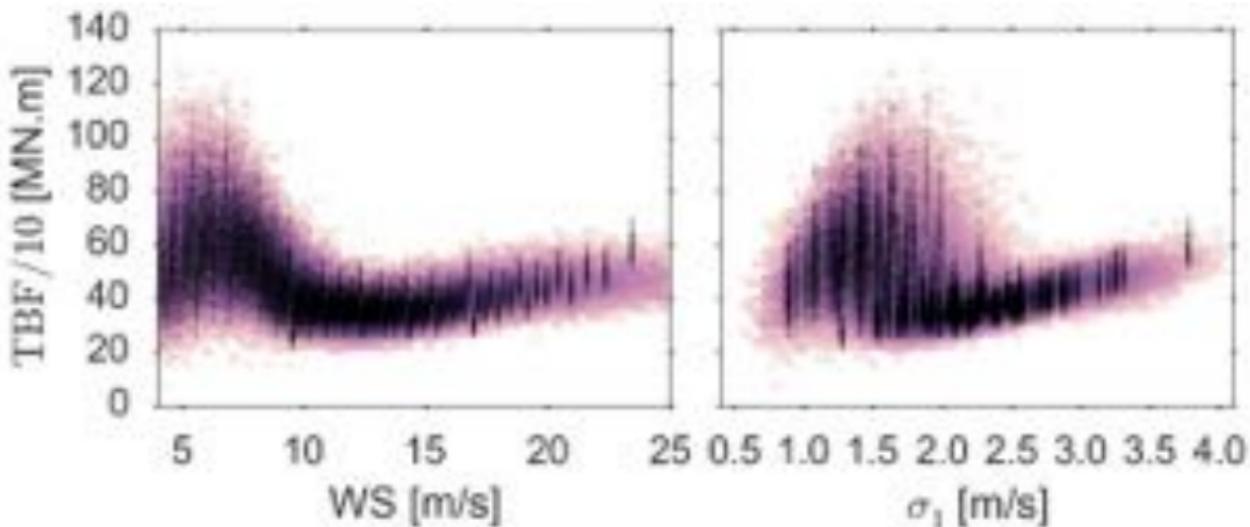
# DTU 10 MW RWT. Blade root flapwise EFL surrogate



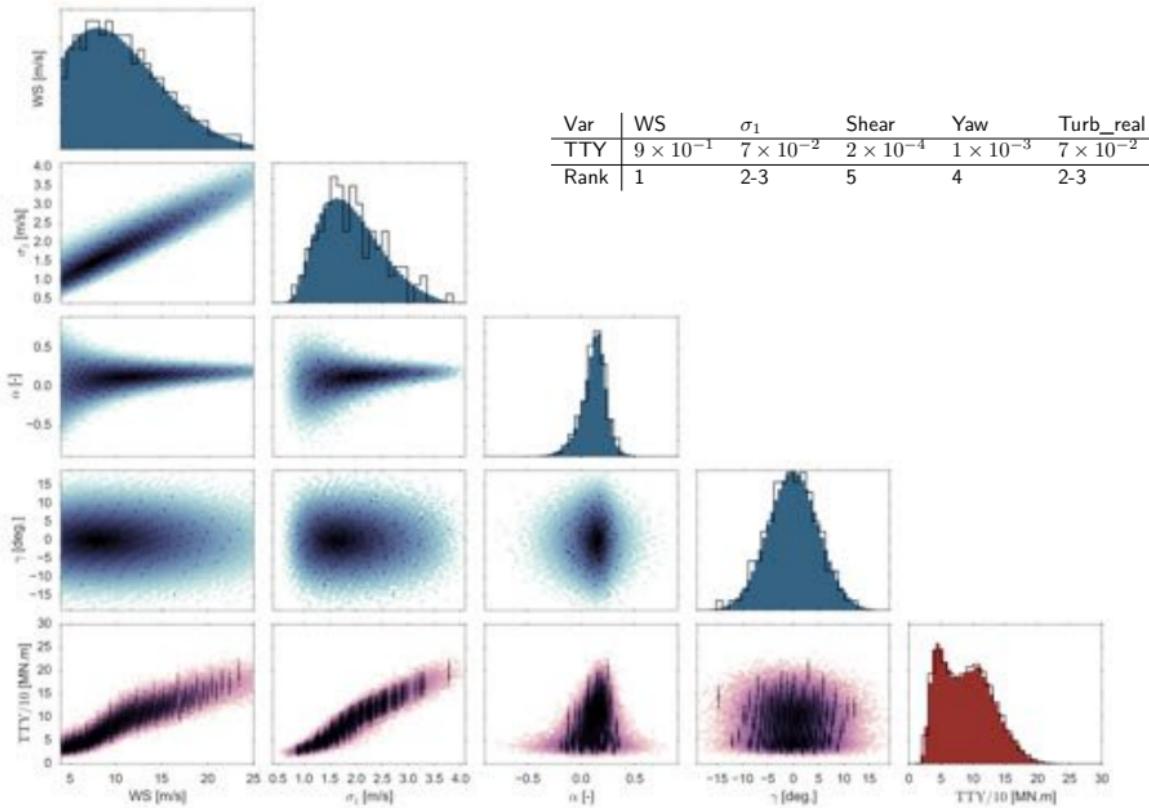
# DTU 10 MW RWT. Tower bottom fore-aft EFL surrogate



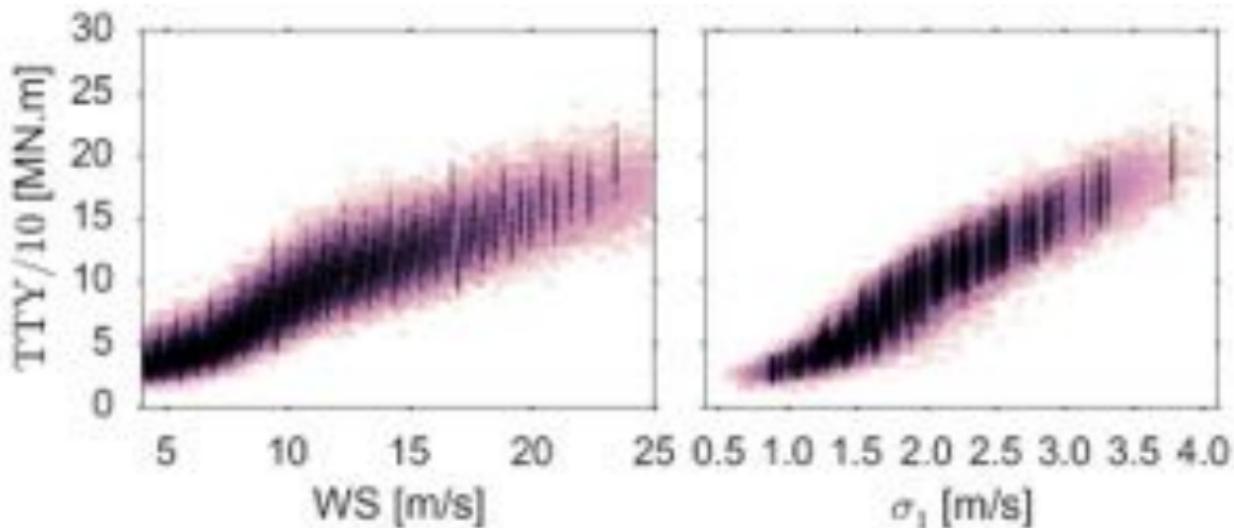
# DTU 10 MW RWT. Tower bottom fore-aft EFL surrogate



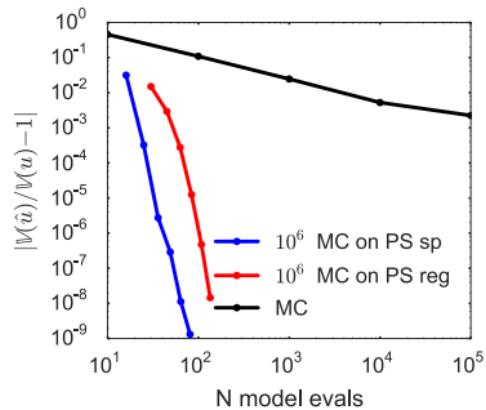
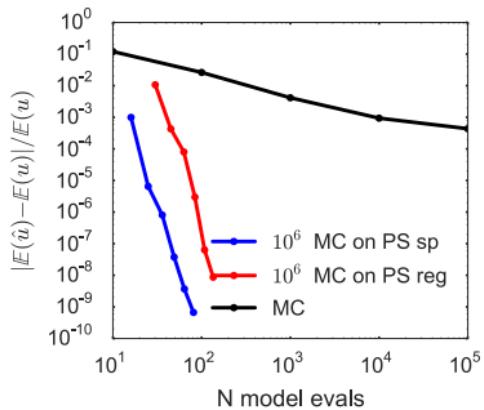
# DTU 10 MW RWT. Tower top yaw EFL surrogate



# DTU 10 MW RWT. Tower bottom fore-aft EFL surrogate



# Convergence example

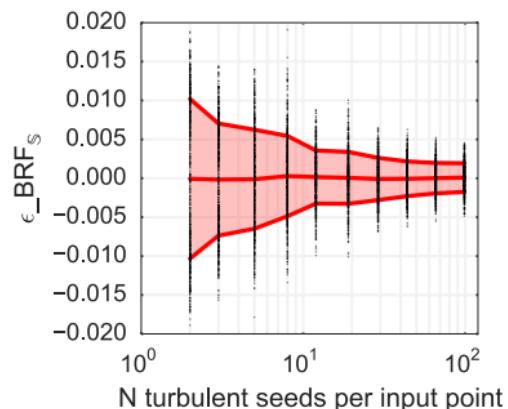
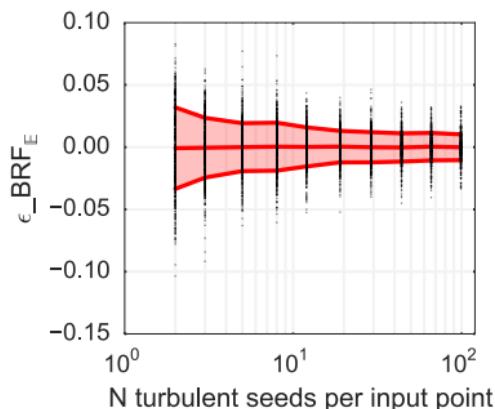


# Uncertainty in the surrogates

Leave-one-out cross validation to estimate the distribution of prediction errors of the surrogates

$$\hat{y}(\mathbf{x}) \sim \text{Normal}(\hat{y}_{\mathbb{E}}(\mathbf{x}) + \epsilon_{y\mathbb{E}} \max(y), \hat{y}_{\mathbb{S}}(\mathbf{x}) + \epsilon_{y\mathbb{S}} \max(y))$$

$$\epsilon_{y\mathbb{E}} = \frac{\hat{y}_{\mathbb{E}}(\mathbf{x}_{LO}) - y_{\mathbb{E}}(\mathbf{x}_{LO})}{\max(y)} \quad \epsilon_{y\mathbb{S}} = \frac{\hat{y}_{\mathbb{S}}(\mathbf{x}_{LO}) - y_{\mathbb{S}}(\mathbf{x}_{LO})}{\max(y)}$$



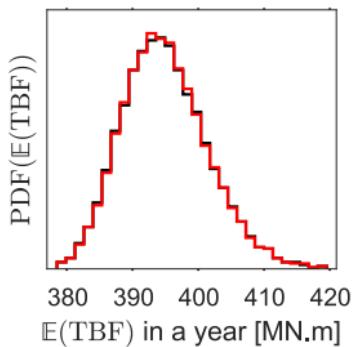
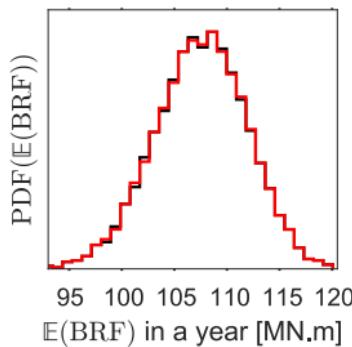
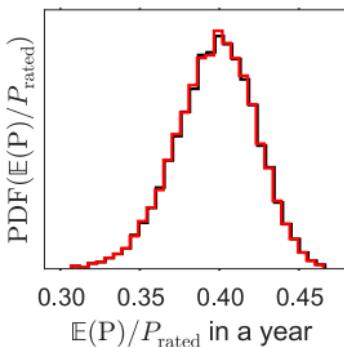
# Uncertainty in AEP and E(EFL).

## Two nested propagations of uncertainty.

Use the surrogate of the DTU 10 MW RWT with uncertain WS resources.

Assume a distribution of the 1 year WS resources

Variable	Distribution	Parameters	
$A$	Normal	$\mu_A = 9$	$\sigma = 0.5 \text{ m/s}$
$k$	Normal	$\mu_A = 2$	$\sigma = 0.1$
$x_0 = \text{WS}$	Weibull	scale = $A$	shape = $k$
$x_1 = \sigma_1$	Lognormal	$\mu_{\sigma_1}(\text{WS})$	$\sigma_{\sigma_1}(\text{WS})$
$x_2 = \alpha$	Normal	$\mu_\alpha(\text{WS})$	$\sigma_\alpha(\text{WS})$
$x_3 = \gamma$	Normal	$\mu_\gamma = 0$	$\sigma_\gamma = 5 \text{ deg.}$



# Conclusions

The surrogates are a way to obtain load estimation under site specific characteristics without sharing the proprietary aero-elastic design.

## PCE as Aero-ealastic model surrogate:

- Efficient uncertainty propagation that enables to compute the statistics of the output such as: mean, standard deviation and sensitivity analysis.
- Effect of turbulent seed requires to estimate the mean and variance for every simulation using a sample of turbulent seeds for each output.
- The surrogate is able to predict both the local and global distribution of the P, CT, EFL.
- It is possible to use the surrogate inside a wind power plant optimization framework and inside uncertainty estimations of AEP and lifetime EFL.

# Future Work

## Surrogates

- NREL 5 MW floating RWT. Under the 5D input case [Graf 2015].
- DTU 10 MW RWT operating inside a wind farm. Dynamic wake meandering model to predict input flow conditions. 9D input case.

## Sensitivity analysis and uncertainty propagation:

- PCE for SA with large number of inputs (more than 100) applied to WAsP.

## Reliability and failure estimation:

- Importance sampling with PCE for extreme loads

# References I

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- [Saltelli 2010] Saltelli, A., Annoni, P., Azzini, I., Campolongo, F., Ratto, M. and Tarantola, S. (2010) Variance based sensitivity analysis of model output. Design and estimator for the total sensitivity index. *Computer Physics Communications*, 181(2), 259–270.

# Questions?

J. P. Murcia  
+45 2339 7790  
[jumu@dtu.dk](mailto:jumu@dtu.dk)  
PhD Student  
DTU Wind Energy

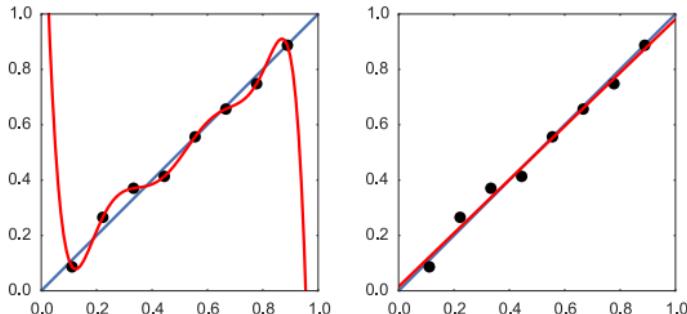
Technical University of Denmark (DTU)  
Building 101  
Risø Campus  
Frederiksborgvej 399  
4000 Roskilde, Denmark

# How to avoid over fitting and achieve sparsity?

## Sparse linear model regression

- Least Absolute Shrinkage and Selection Operator problem (LASSO) is useful to avoid over-fitting and achieve sparsity in the PCE.
- LASSO is a least squares minimization problem with a  $l_1$  penalization on the coefficients ( $c$ ):

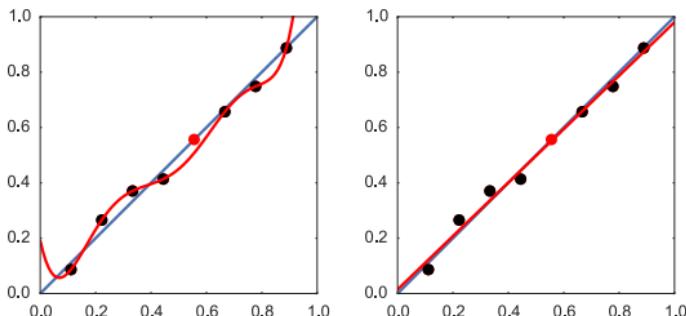
$$\min_{\mathbf{c}} \sum_{i=0}^{N-1} \left[ \sum_{l=0}^{N_c-1} c_l \phi_l(\mathbf{w}_i) - y(\mathbf{x}_i) \right]^2 + \alpha \sum_{l=0}^{N_c-1} |c_l|$$



# How to select the right sparsity?

## k-fold cross validation

- It divides the dataset in  $k$  groups and uses  $k-1$  groups ("folds") for training and the remaining for validation. Repeat this process until all the groups have been the validation set.
- k-fold cross validation is repeated for multiple values of the sparsity parameter ( $\alpha$ ).
- As a result it gives the optimal sparsity parameter ( $\alpha$ ).



# DTU 10 MW RWT.

## 20-fold Cross validation examples

- Each blue line is the mean square error of a single k-fold prediction (trained with 95% of the data and tested in the remaining 5%)
- Black line is the average over all the validation set combinations

